

CHAPTER 4. THE ENERGY EQUATION

1. The energy equation
2. The mechanical energy equation. Definition of energy loss and headloss
3. Bernoulli equation in steady-state. Piezometric head constancy
4. Concepts related to power and efficiency

1. The energy equation

1. General equation of conservation of energy*
2. Simplified forms of the energy equation*
3. Average velocity – Coriolis coefficient*
4. Application to a streamline*

1.1. General equation of conservation of energy

✓ Word statement:

$$\left\{ \begin{array}{l} \text{Rate of energy} \\ \text{transfer to the system} \end{array} \right\} = \left\{ \begin{array}{l} \text{Rate of energy change} \\ \text{(increase) in the system} \end{array} \right\}$$

- Intrinsic energy:

- Kinetic energy
- Potential energy
- Internal energy
- Chemical energy
- Nuclear energy

- Energy transfer:

- Heat
 - Work
- } Signs agreement

1.1. General equation of conservation of energy

✓ Formulation:

$$\frac{dQ}{dt} - \frac{dW}{dt} = \frac{dE_{sys}}{dt}$$

• Intrinsic energy:

- Kinetic energy
- Potential energy
- Internal energy



$$b = e = \tilde{u} + \frac{U^2}{2} + gz$$

$$E_{sys} = \iiint_{V_{sys}} \rho \left(\tilde{u} + \frac{U^2}{2} + gz \right) dV$$

• Transport theorem:

$$\frac{dB_{sys}}{dt} = \frac{d}{dt} \iiint_{V_c} \rho b dV + \oiint_{A_c} b \rho \vec{U} d\vec{A}$$

$$\dot{Q} - \dot{W} = \frac{d}{dt} \iiint_{V_c} \rho \left(\tilde{u} + \frac{U^2}{2} + gz \right) dV + \oiint_{A_c} \rho \left(\tilde{u} + \frac{U^2}{2} + gz \right) \vec{U}_r d\vec{A}$$

1.1. General equation of conservation of energy

- ✓ Analysis of the types of heat (Course: Thermal technology in the 3rd year)
 - Conduction
 - Convection
 - Radiation

- ✓ Analysis of the types of work
 - Shaft work
 - Work done by shear-stresses at the control surface
 - Work done by pressure at the control surface:
 - Flux work
 - Deformation work

1.1. General equation of conservation of energy

✓ Analysis of the types of works

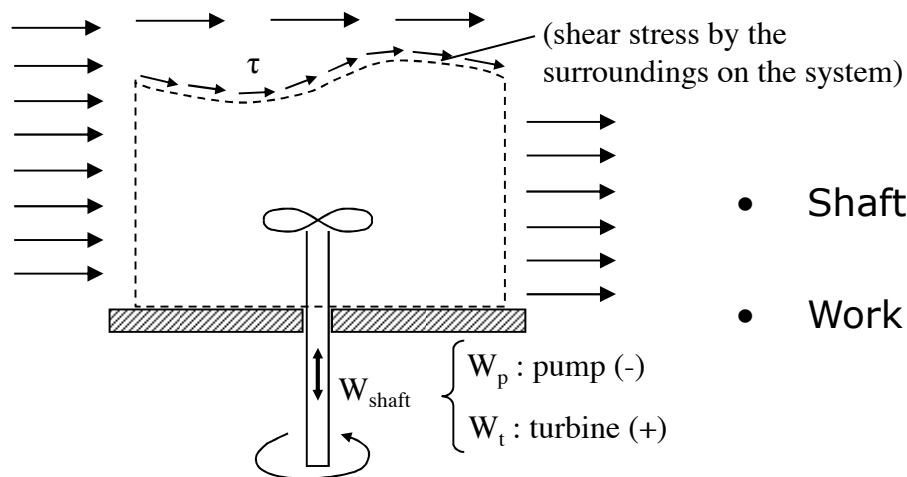


Figure 4.1 Analysis of the types of mechanical work

- Shaft work: $\dot{W}_{\text{shaft}} = M \cdot \omega$
- Work done by shear stresses:

$$\dot{W}_{\text{shear}} = -\iint_{A_c} \vec{U}(\vec{\tau}dA)$$

- Work done by pressure:

- Flux work:

$$\dot{W}_{\text{flux}} = \iint_{A_c} \vec{U}_r p d\vec{A}$$

- Deformation work:

$$\frac{dW_D}{dt} = \dot{W}_D = \iint_{A_c} \vec{U}_c p d\vec{A}$$

1.1. General equation of conservation of energy

✓ Energy equation

$$\dot{Q} - \dot{W} = \frac{d}{dt} \iiint_{V_c} \rho \left(\tilde{u} + \frac{U^2}{2} + gz \right) dV + \iint_{A_c} \rho \left(\tilde{u} + \frac{U^2}{2} + gz \right) \tilde{U}_r d\vec{A}$$



$$\dot{Q} - (\dot{W}_{\text{shaft}} + \dot{W}_{\text{shear}}) - \iint_{A_c} p \tilde{U}_r d\vec{A} - \iint_{A_c} p \tilde{U}_c d\vec{A} = \frac{d}{dt} \iiint_{V_c} \rho \left(\tilde{u} + \frac{U^2}{2} + gz \right) dV + \iint_{A_c} \rho \left(\tilde{u} + \frac{U^2}{2} + gz \right) \tilde{U}_r d\vec{A}$$



$$\dot{Q} - (\dot{W}_{\text{shaft}} + \dot{W}_{\text{shear}}) - \iint_{A_c} p \tilde{U}_c d\vec{A} = \frac{d}{dt} \iiint_{V_c} \rho \left(\tilde{u} + \frac{U^2}{2} + gz \right) dV + \iint_{A_c} \rho \left(\tilde{u} + \frac{p}{\rho} + \frac{U^2}{2} + gz \right) \tilde{U}_r d\vec{A}$$

“Integral approach of the General Equation of Energy”

1.1. General equation of conservation of energy

- ✓ Energy equation

$$\dot{Q} - (\dot{W}_{\text{shaft}} + \dot{W}_{\text{shear}}) - \oint_{A_c} p \bar{U}_c d\bar{A} = \frac{d}{dt} \iiint_{V_c} \rho \left(\tilde{u} + \frac{U^2}{2} + gz \right) dV + \oint_{A_c} \rho \left(\tilde{u} + \frac{p}{\rho} + \frac{U^2}{2} + gz \right) \bar{U}_r d\bar{A}$$

- ✓ Steady-state regime and rigid and fixed control volume:

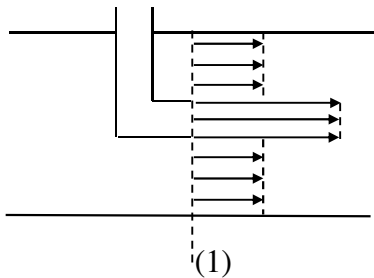
$$\dot{Q} - (\dot{W}_{\text{shaft}} + \dot{W}_{\text{shear}}) = \oint_{A_c} \rho \left(\tilde{u} + \frac{p}{\rho} + \frac{U^2}{2} + gz \right) \bar{U} d\bar{A}$$

- ✓ Finite number of inlet and outlet, uniform flow:

$$\dot{Q} - (\dot{W}_{\text{shaft}} + \dot{W}_{\text{shear}}) = \sum_{\text{out}} q_m \left(\tilde{u} + \frac{p}{\rho} + \frac{U^2}{2} + gz \right) - \sum_{\text{in}} q_m \left(\tilde{u} + \frac{p}{\rho} + \frac{U^2}{2} + gz \right)$$

1.3. Average velocity – Coriolis' coefficient

- ✓ Average velocities and properties
- ✓ Kinetic energy correction coefficient or Coriolis' coefficient α



$$\dot{E}_k = \iint_A \rho \frac{U^3}{2} dA \neq \rho \frac{U_{av}^3}{2} A$$

$$\dot{E}_k = \alpha \cdot \rho \frac{U_{av}^3}{2} A = \alpha \cdot q_m \frac{U_{av}^2}{2}$$

$$\alpha = \frac{\iint_A U^3 dA}{U_{av}^3 A}$$

Figure 4.2 Jet injection inside a pipe

- ✓ Finite number of inlets and outlets:

$$\dot{Q} - (\dot{W}_{shaft} + \dot{W}_{shear}) = \sum_{out} q_m \left(\tilde{u} + \frac{p}{\rho} + \alpha \frac{U_{av}^2}{2} + gz \right) - \sum_{in} q_m \left(\tilde{u} + \frac{p}{\rho} + \alpha \frac{U_{av}^2}{2} + gz \right)$$

1.4. Application to a streamline

- ✓ Energy equation in a streamline

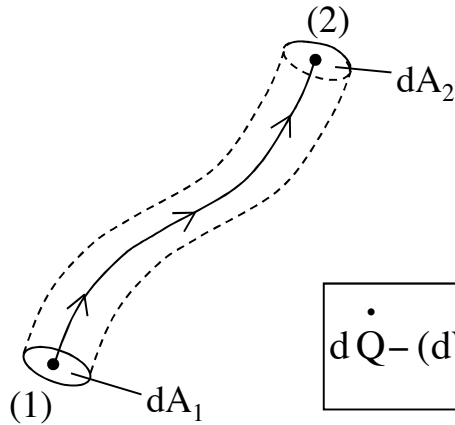
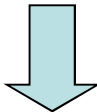


Figure 4.3 Streamline and differential streamtube

$$d\dot{Q} - (d\dot{W}_{\text{shaft}} + d\dot{W}_{\text{shear}}) = dq_{m2} \left(\tilde{u}_2 + \frac{p_2}{\rho_2} + \frac{U_2^2}{2} + gz_2 \right) - dq_{m1} \left(\tilde{u}_1 + \frac{p_1}{\rho_1} + \frac{U_1^2}{2} + gz_1 \right)$$


 Specific properties: $q = \frac{d\dot{Q}}{dq_m}$ $w_{\text{shaft}} = \frac{d\dot{W}_{\text{shaft}}}{dq_m}$ $w_{\text{shear}} = \frac{d\dot{W}_{\text{shear}}}{dq_m}$

$$q - (w_{\text{shaft}} + w_{\text{shear}}) = (\tilde{u}_2 - \tilde{u}_1) + \left(\frac{p_2}{\rho_2} - \frac{p_1}{\rho_1} \right) + \left(\frac{U_2^2}{2} - \frac{U_1^2}{2} \right) + (gz_2 - gz_1)$$

$$\left(\tilde{u}_1 + \frac{p_1}{\rho_1} + \frac{U_1^2}{2} + gz_1 \right) + q - (w_{\text{shaft}} + w_{\text{shear}}) = \left(\tilde{u}_2 + \frac{p_2}{\rho_2} + \frac{U_2^2}{2} + gz_2 \right)$$

2. The mechanical energy equation. Definition of energy loss and headloss

✓ The mechanical energy equation

$$q - (w_{\text{shaft}} + w_{\text{shear}}) = (\tilde{u}_2 - \tilde{u}_1) + \left(\frac{p_2}{\rho_2} - \frac{p_1}{\rho_1} \right) + \left(\frac{U_2^2}{2} - \frac{U_1^2}{2} \right) + (gz_2 - gz_1)$$

$$-(w_{\text{shaft}} + w_{\text{shear}}) - \left[(\tilde{u}_2 - \tilde{u}_1) + \int_1^2 p dv - q \right] - \int_1^2 \frac{dp}{\rho} = \left(\frac{U_2^2}{2} - \frac{U_1^2}{2} \right) + (gz_2 - gz_1)$$

$$-(w_{\text{shaft}} + w_{\text{shear}}) - \int_1^2 \frac{dp}{\rho} - (gh_L) = \left(\frac{U_2^2}{2} + gz_2 \right) - \left(\frac{U_1^2}{2} + gz_1 \right)$$

Work to machinery
or shear stresses

Work by
pressure
forces

Mechanical
energy loss

Change of specific mechanical
energy of the fluid

Mechanical energy equation
for a steady-state flow along
a streamline

$$(gh_L) = \left[(\tilde{u}_2 - \tilde{u}_1) + \int_1^2 p dv - q \right]$$

Mechanical energy loss

2. The mechanical energy equation. Definition of energy loss and headloss

- ✓ Mechanical energy equation. Incompressible flow

$$\frac{p_1}{\rho} + \frac{U_1^2}{2} + gz_1 - (w_{\text{shaft}} + w_{\text{shear}}) = \frac{p_2}{\rho} + \frac{U_2^2}{2} + gz_2 + gh_L$$

$$(gh_L) = [(\tilde{u}_2 - \tilde{u}_1) - q]$$

The diagram illustrates the decomposition of the head loss term (gh_L) from the mechanical energy equation. The equation $(gh_L) = [(\tilde{u}_2 - \tilde{u}_1) - q]$ is shown at the top. Below it, two curly braces are positioned under the terms $(\tilde{u}_2 - \tilde{u}_1)$ and $-q$. Lines connect these braces to two larger curly braces below. The first brace is labeled "Gain of internal energy" and the second is labeled "Heat given to the surroundings". A plus sign is placed between these two braces, indicating that the head loss is the sum of the gain in internal energy and the heat lost to the surroundings.

$$\left\{ \begin{array}{c} \text{Gain of internal} \\ \text{energy} \end{array} \right\} + \left\{ \begin{array}{c} \text{Heat given to the} \\ \text{surroundings} \end{array} \right\}$$

3. Bernoulli equation

- ✓ Bernoulli equation in steady-state regime

$$-(w_{\text{shaft}} + w_{\text{shear}}) - \int_1^2 \frac{dp}{\rho} - (gh_L) = \left(\frac{U_2^2}{2} + gz_2 \right) - \left(\frac{U_1^2}{2} + gz_1 \right)$$



- ✓ Flow without any friction (inviscid fluid)
- ✓ Flow without any heat transfer (neither addition nor extraction)
- ✓ Flow without shaft work between (1) and (2)

$$\int_1^2 \frac{dp}{\rho} + \frac{1}{2} (U_2^2 - U_1^2) + g(z_2 - z_1) = 0$$

"Bernoulli equation in a compressible fluid" (steady-state regime) valid for a streamline.

- Incompressible fluid

$$gz + \frac{p}{\rho} + \frac{U^2}{2} = \text{Cte}$$

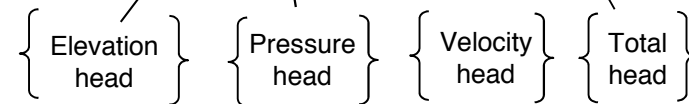
3. Bernoulli equation

- ✓ Alternative forms of Bernoulli equation

$$gz + \frac{p}{\rho} + \frac{U^2}{2} = \text{Cte}$$

$$\gamma z + p + \frac{1}{2} \rho U^2 = \text{Cte} = p_T$$

$$z + \frac{p}{\gamma} + \frac{U^2}{2g} = \text{Cte} = h_T$$



- ✓ Energy heads diagram

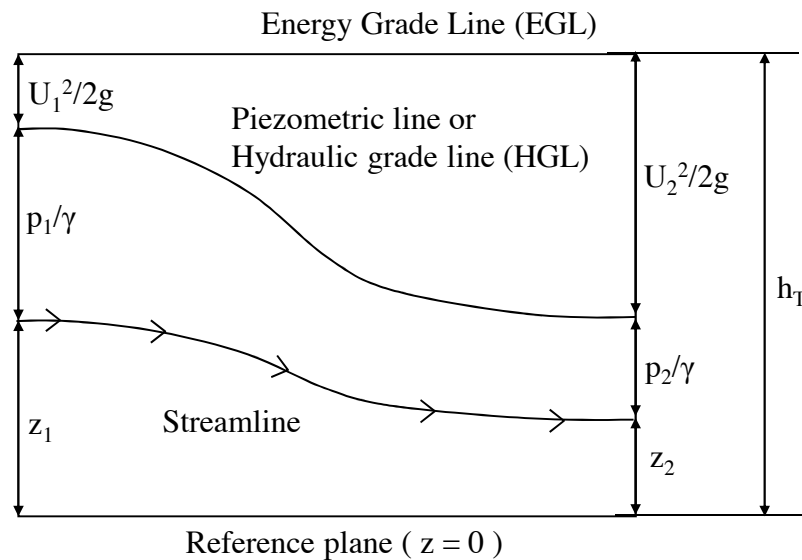


Figure 4.4 Energy heads diagram

3. Piezometric head constancy

✓ Piezometric head constancy

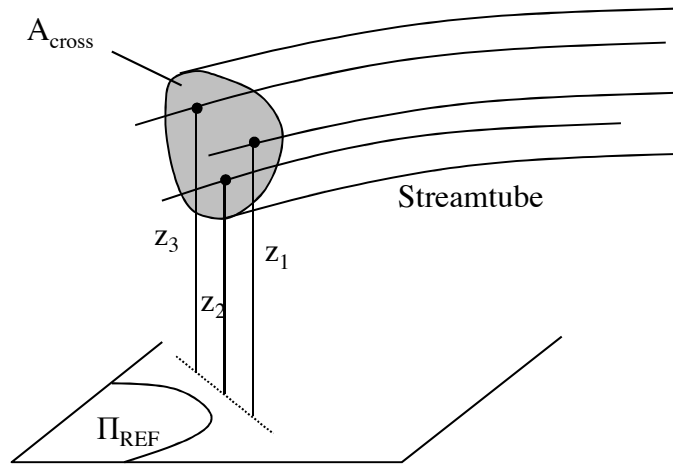


Figure 4.5 Piezometric head constancy

$$\frac{p_1}{\gamma} + z_1 = \frac{p_2}{\gamma} + z_2 = \frac{p_3}{\gamma} + z_3 = Cte$$

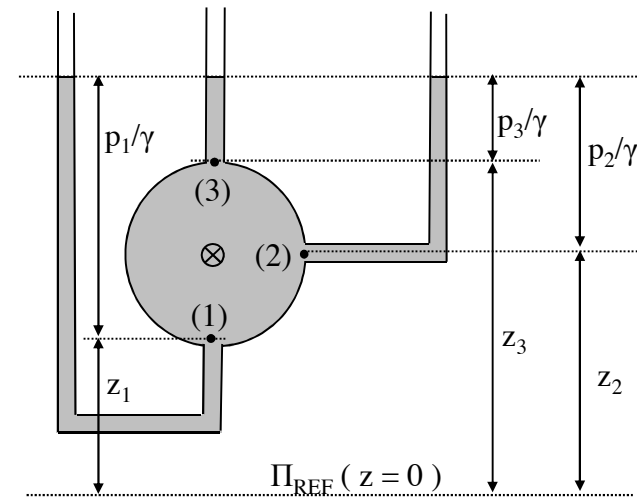


Figure 4.6 Manometers connected to a pipe

3. Bernoulli equation

✓ Generalization to a streamtube of:

- Bernoulli equation

$$z_1 + \frac{p_1}{\gamma} + \alpha_1 \frac{U_1^2}{2g} = z_2 + \frac{p_2}{\gamma} + \alpha_2 \frac{U_2^2}{2g} = h_{T, \text{section}}$$

- Steady-state mechanical energy equation

$$z_1 + \frac{p_1}{\gamma} + \alpha_1 \frac{U_1^2}{2g} + H_p = z_2 + \frac{p_2}{\gamma} + \alpha_2 \frac{U_2^2}{2g} + H_t + h_L$$

{ Net head of the pump } { Net head of the turbine } { Headloss }

4. Definition of power terms

- ✓ Power of a finite stream

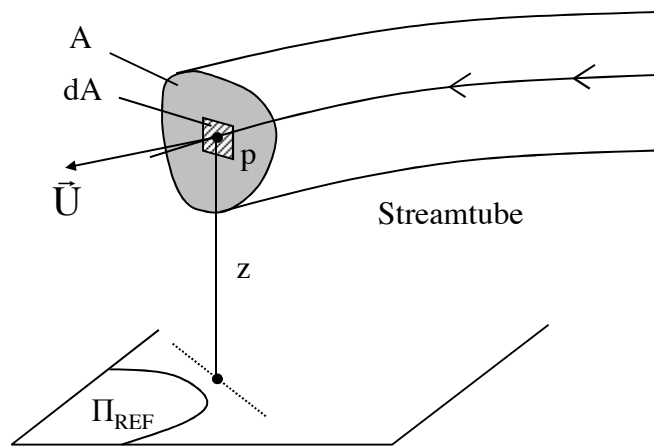


Figure 4.8 Streamtube

$$h_T = z + \frac{p}{\gamma} + \frac{U^2}{2g}$$



$$\dot{E} = \gamma Q \left(z + \frac{p}{\gamma} + \alpha \frac{U_{av}^2}{2g} \right) = \gamma Q h_{T,section}$$

4. Definition of power terms

- ✓ Dissipated power

$$\dot{E}_{\text{lost}} = \gamma Q h_L$$

- ✓ Power of a pump

$$\left| \dot{W}_p \right| = \gamma Q H_p$$

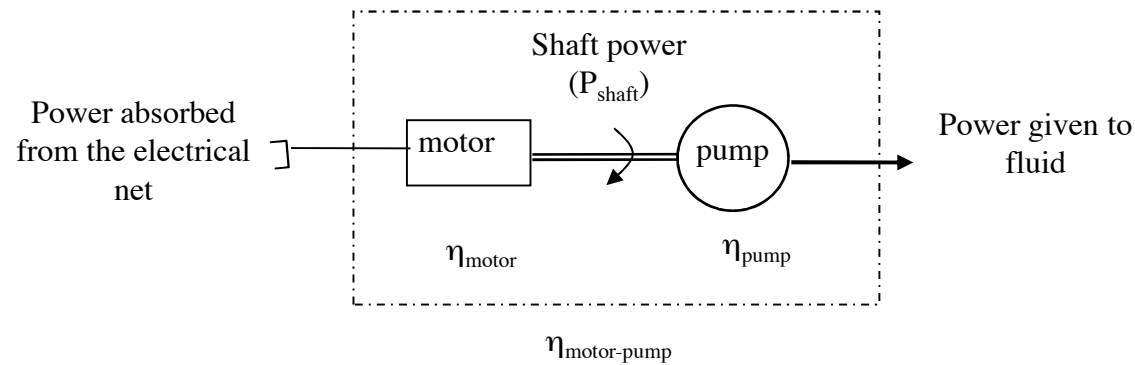
- ✓ Power of a turbine

$$\dot{W}_t = \gamma Q H_t$$

4. Definition of power terms

✓ Efficiencies

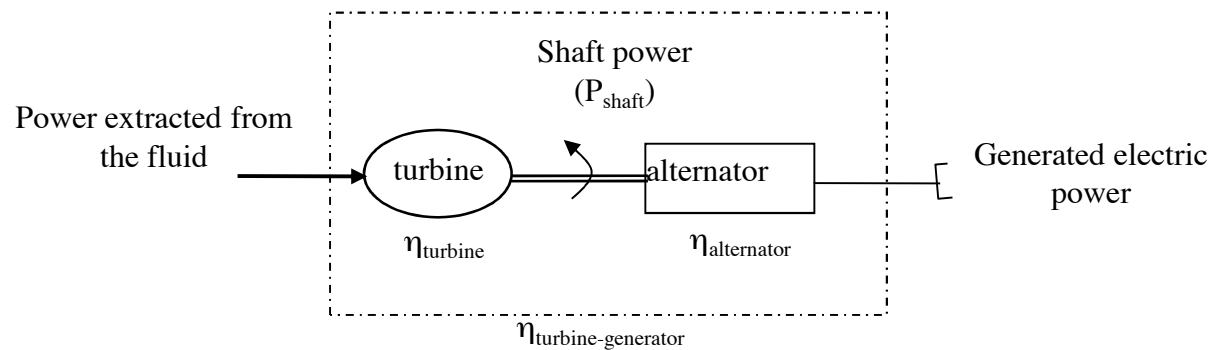
- Pump



$$\eta_{pump} = \frac{|\dot{W}_p|}{P_{shaft}} = \frac{\gamma Q H_p}{P_{shaft}}$$

$$\eta_{motor} = \frac{P_{shaft}}{P_{electric}}$$

- Turbine



$$\eta_{turbine} = \frac{P_{shaft}}{\dot{W}_t} = \frac{P_{shaft}}{\gamma Q H_t}$$

$$\eta_{alternator} = \frac{P_{electric}}{P_{shaft}}$$